

# Precise orbit determination for the CryoSat-2 mission

status of CNES GDR solutions



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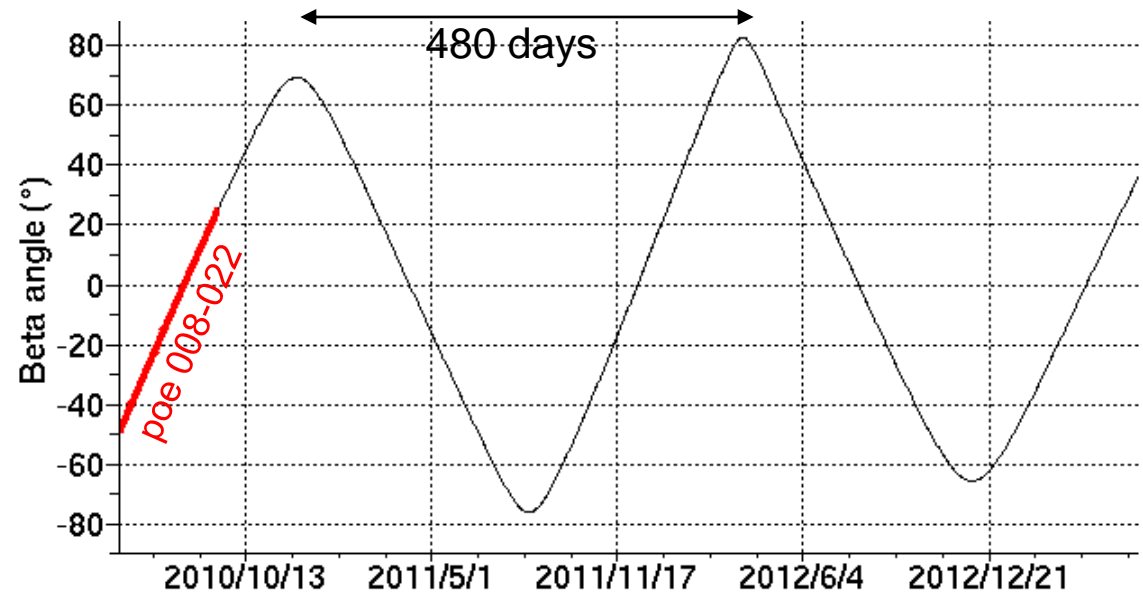
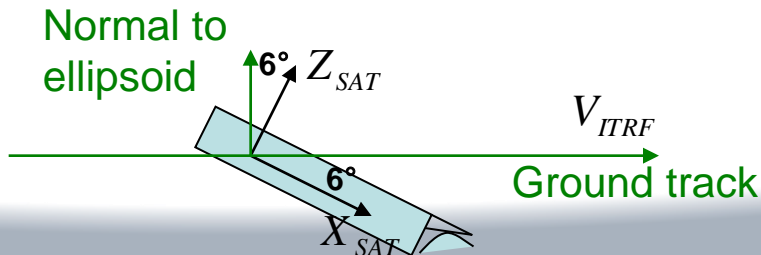
# Contents

- Orbit geometry and attitude model
- Evaluation of orbit accuracy
- Modeling of Atmospheric Drag with strong solar activity
- Modeling of solar radiation pressure
- Impact of ITRF2008 on Cryosat-2 orbit
- Conclusions



# Orbit geometry and attitude law

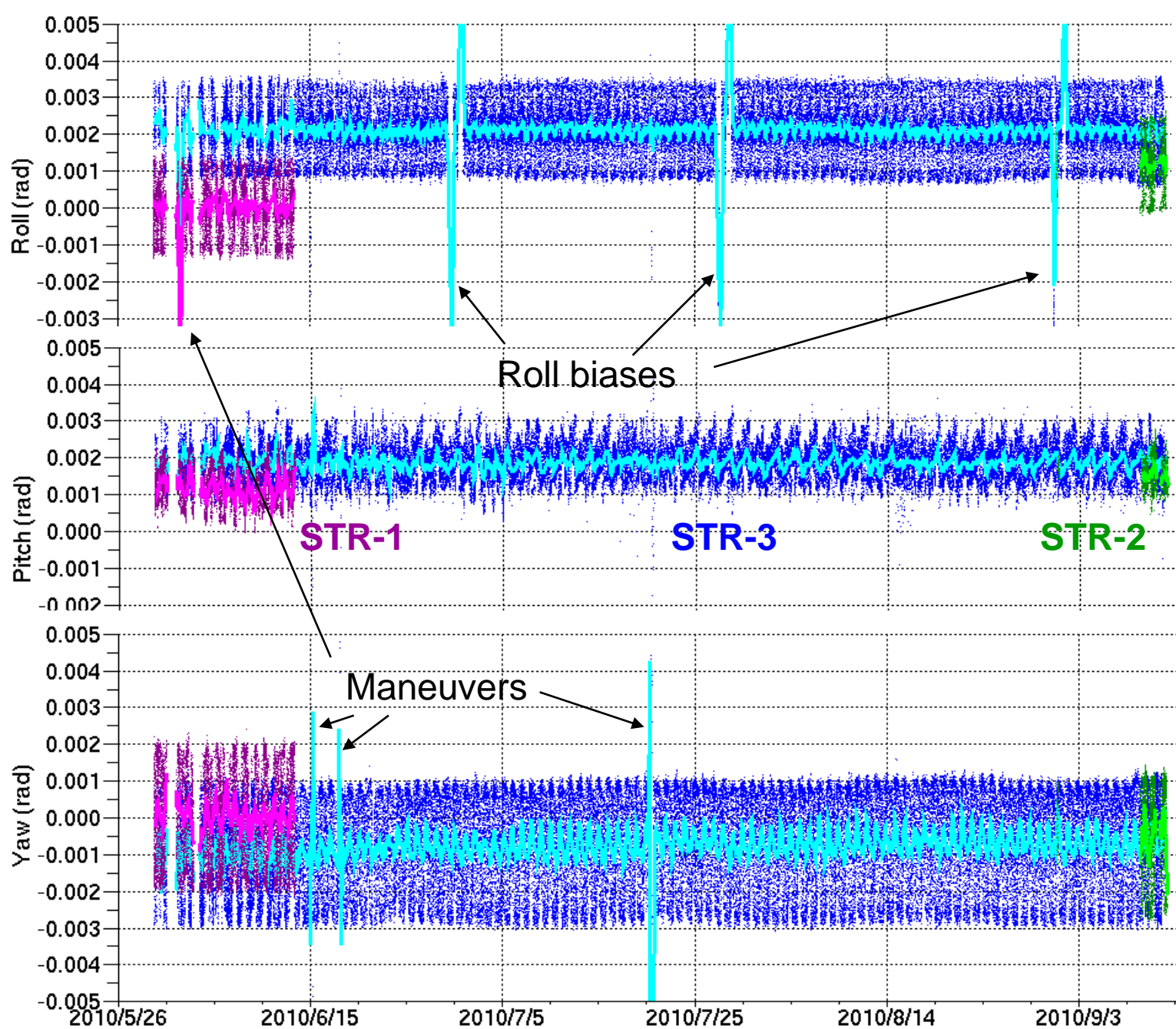
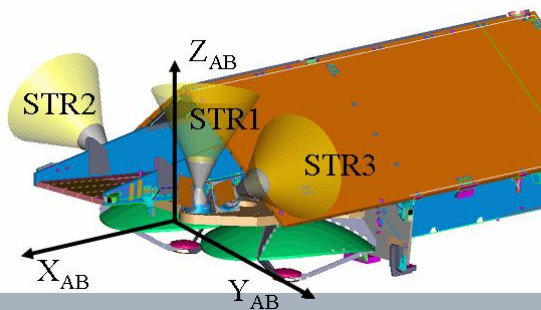
- 92° inclination, 720 km altitude
- Beta angle has a 480-day period
- Attitude law
  - ◆ Satellite Z axis normal to ellipsoid
  - ◆ X towards V in Earth-fixed frame



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[...]

- Nominal attitude is normally consistent with STR data to within  $0.2^\circ$
- Sufficient for POD purposes, but systematic errors could be further reduced



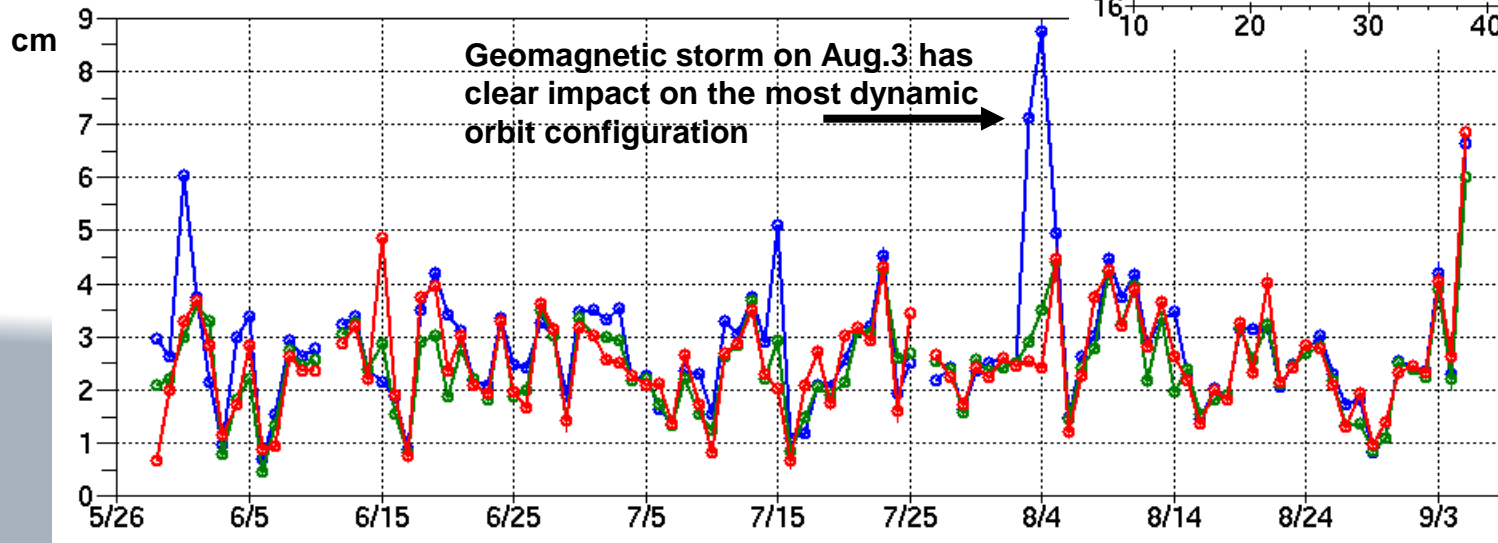
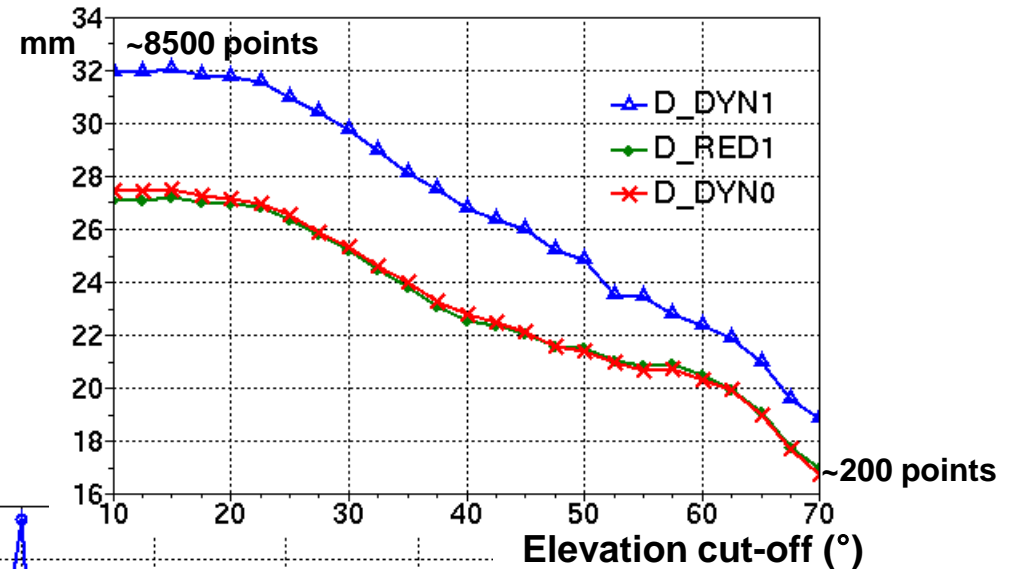
POE ARCS 008→022

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# SLR Residuals on different Doris-only orbits

- High elevation SLR residuals indicate that the doris-only orbit radial accuracy is below 2 cm (similar results are obtained on Envisat Doris-only orbits)

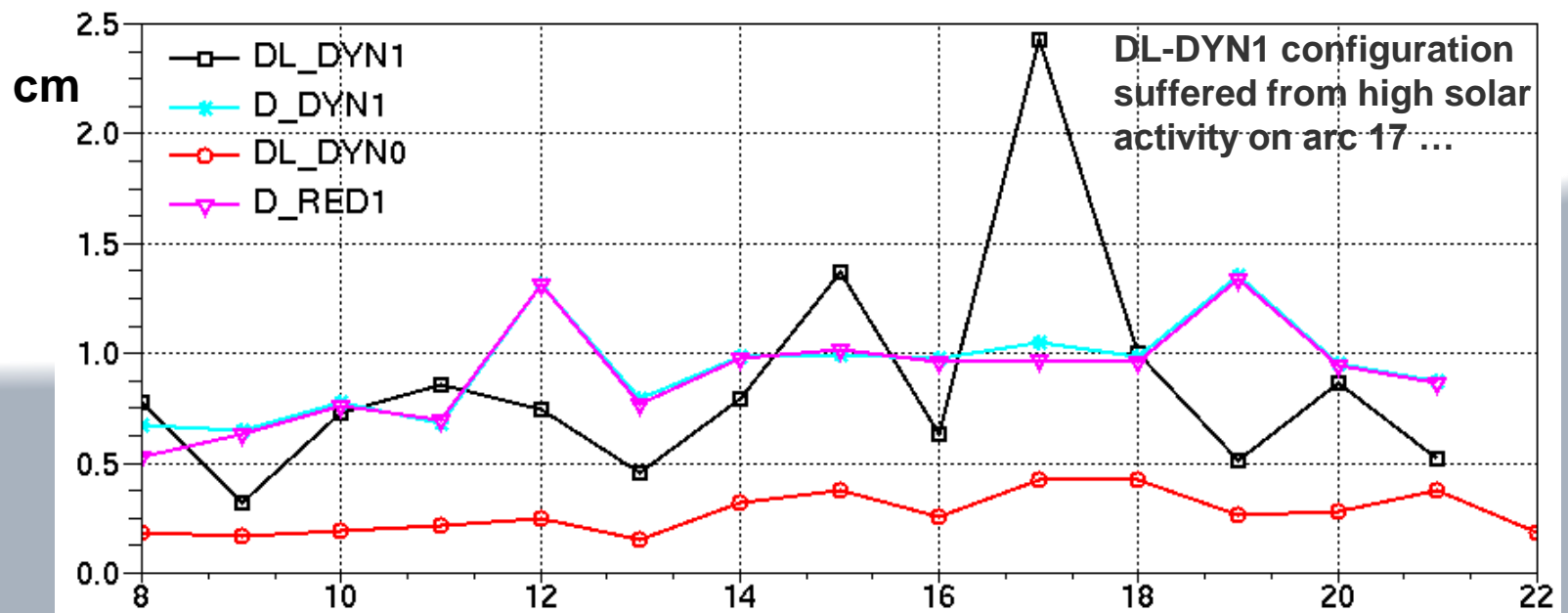
Residuals over arcs 008-022 from YARR\_L7090, WASH\_L7105, MONU\_L7110, GRAZ\_L7839, FORT\_L7080, HERS\_L7840, ZIMM\_L7810



# Radial comparison of different internal solutions

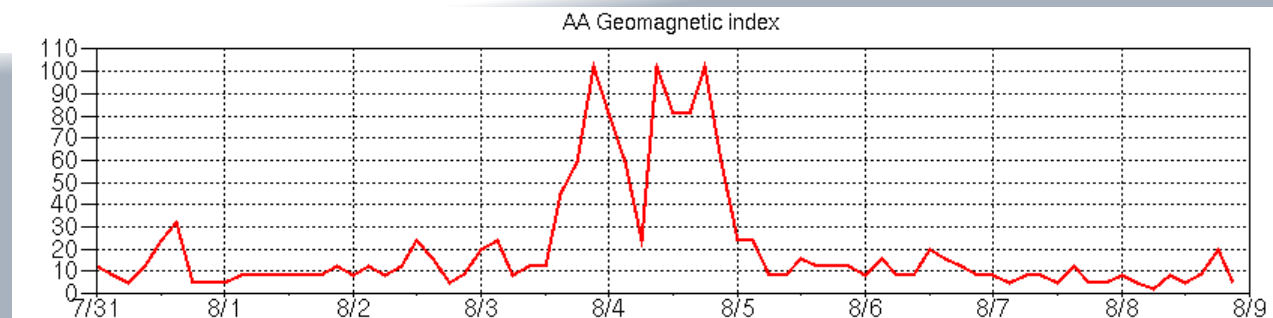
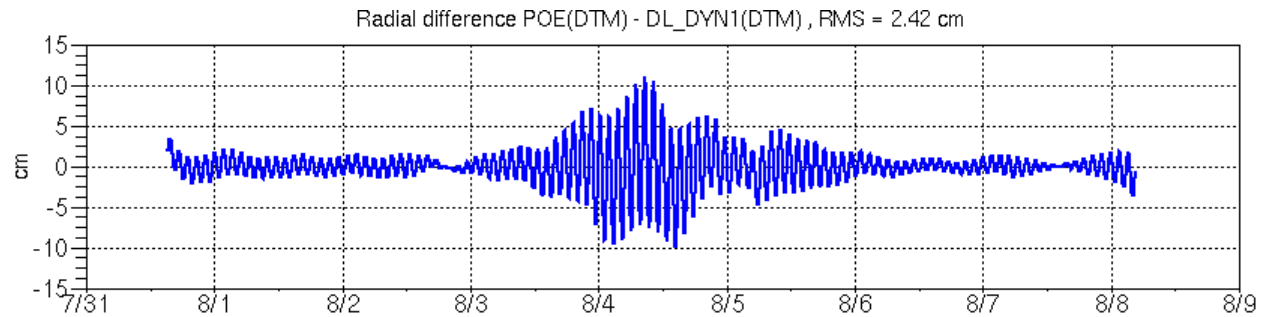
## ■ Comparison of different internal solutions to POE at the 1-cm level

- ◆ POE: Dynamic step (1 drag every 3 revs + 1/revs every 24 hours) + stochastic constant along-track
- ◆ DYN0: POE Dynamic step
- ◆ DYN1: 1 constant along track every 8 hours + 1/revs every 24 hours
- ◆ RED1: DYN1 + stoch. constant along track + stoch. 1/rev along track



# Impact of the atmospheric density model on arc 017

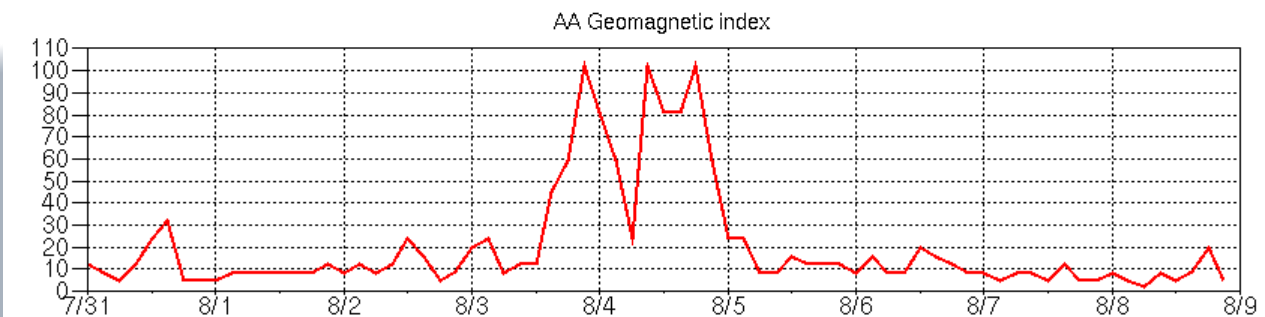
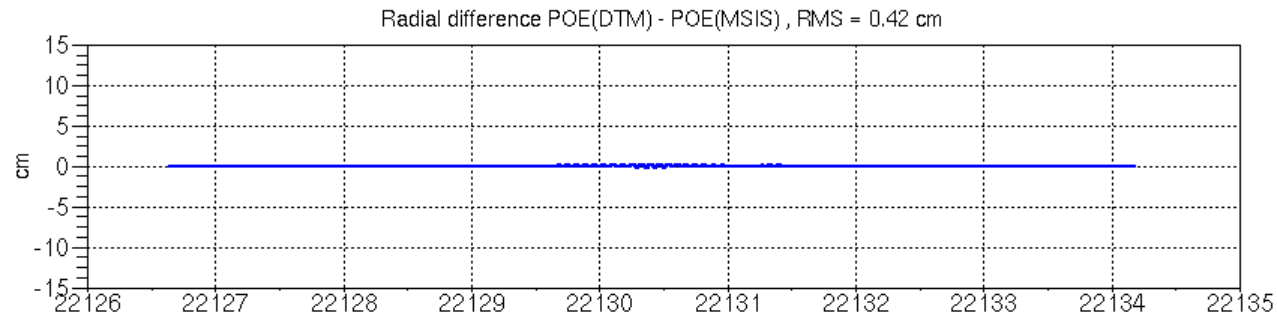
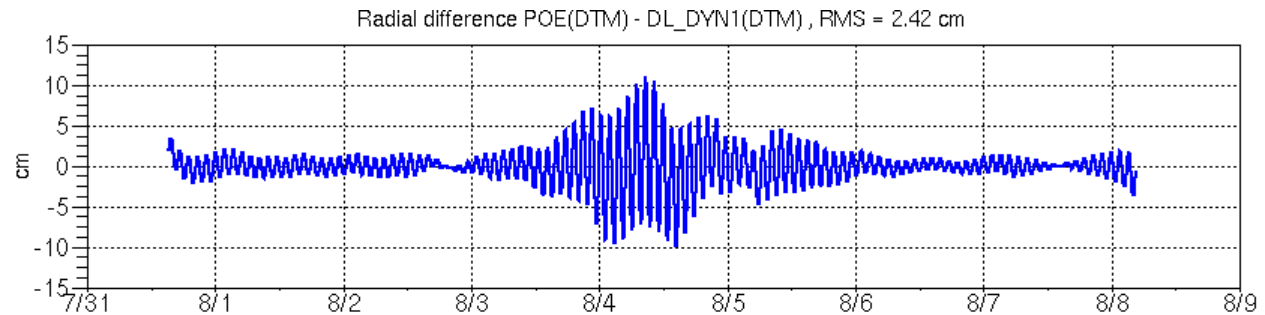
- Use of DTM94 strongly degrades the orbit with a dynamic parameterization



# Impact of the atmospheric density model on arc 017

- Use of DTM94 strongly degrades the orbit with a dynamic parameterization

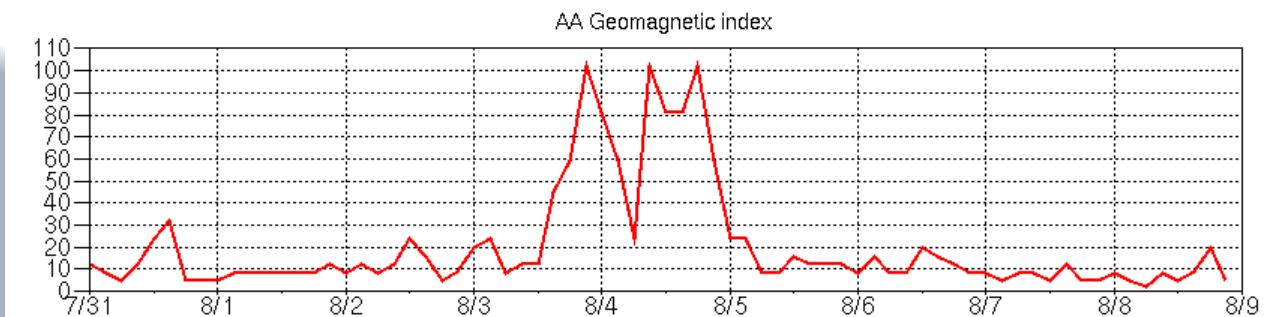
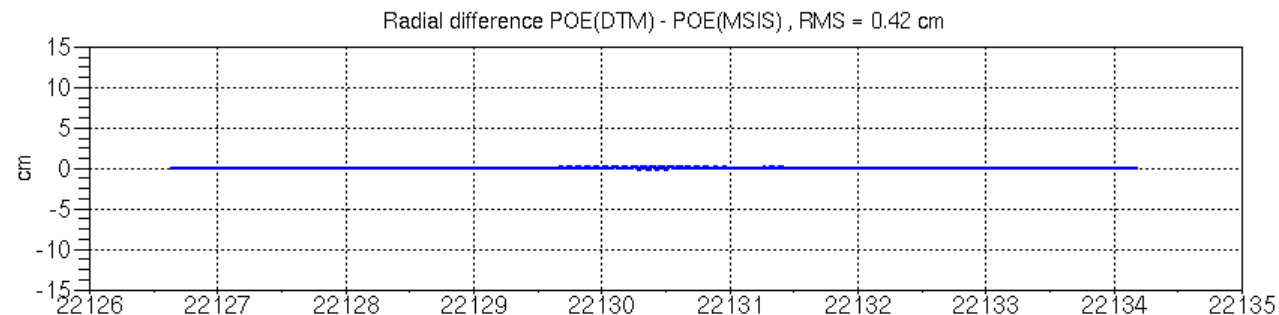
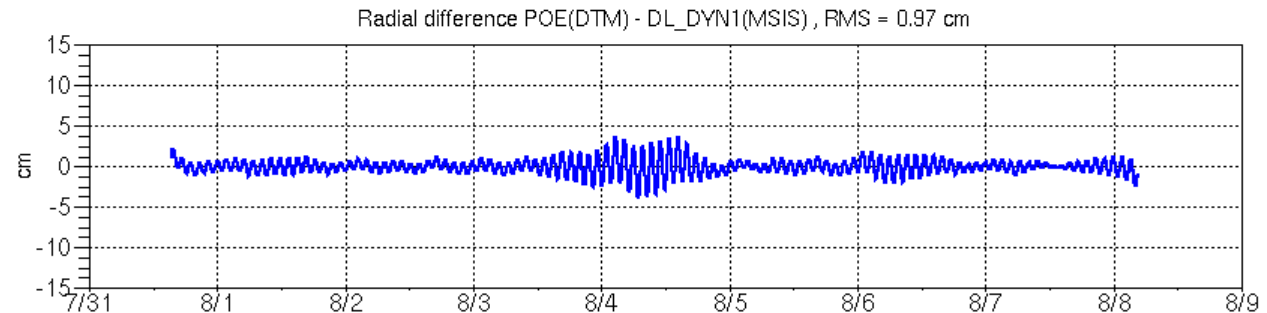
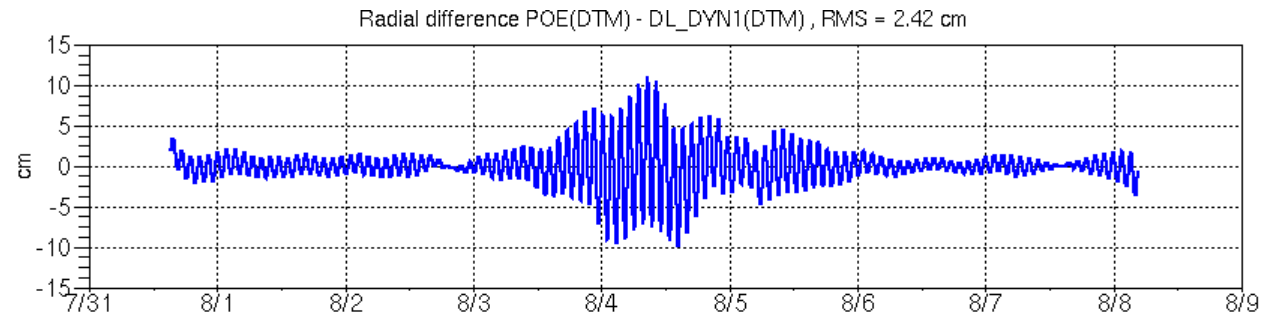
- POE parameterization is not very sensitive to atmospheric density model





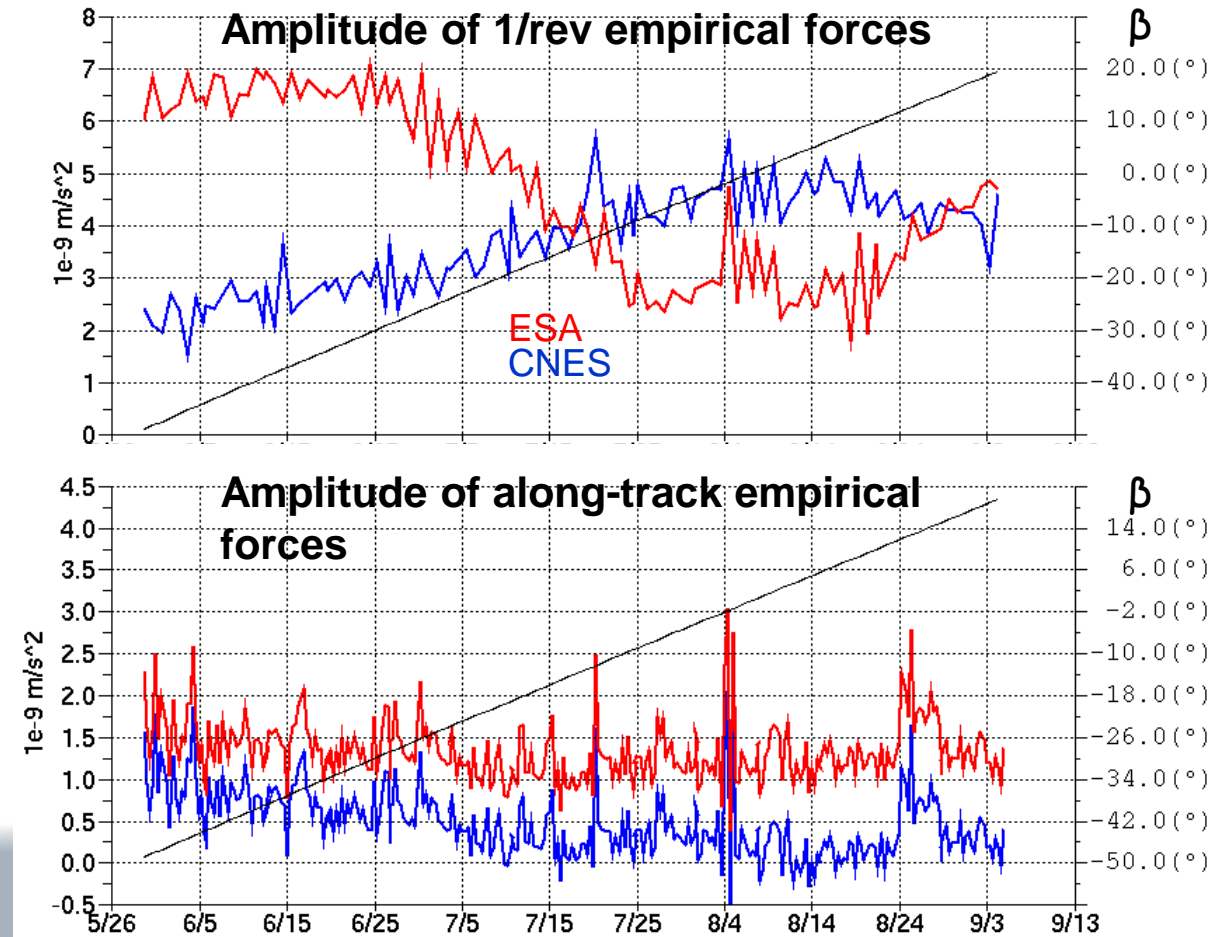
# Impact of the atmospheric density model on arc 017

- Use of DTM94 strongly degrades the orbit with a dynamic parameterization
- MSIS86 allows to improve significantly the dynamic solution
- POE parameterization is not very sensitive to atmospheric density model



# Modeling of solar radiation pressure

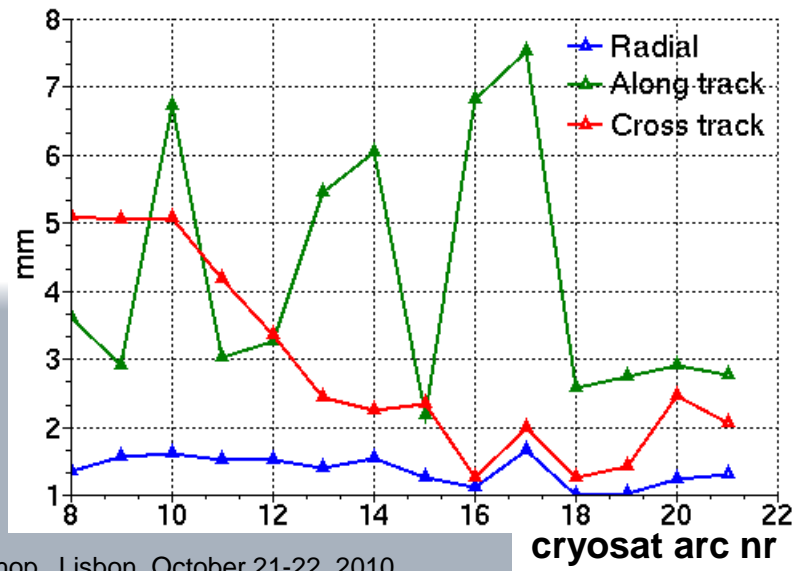
- GDR orbits currently use ESA provided model (6-plates)
- CNES produced an independent model (7-plates) based on satellite photographs, drawings, and typical optical properties of external surfaces



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# Modeling of solar radiation pressure

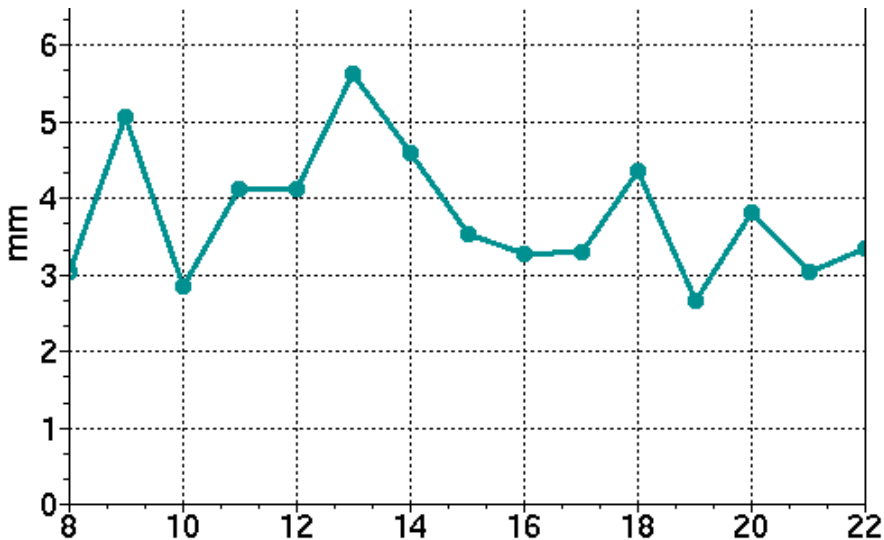
- Amplitude of empirical forces indicates that CNES model performs better when the sun is out of plane, and worse when the sun is in plane
- Possible explanation: a-priori optical properties used by CNES for the solar array can be improved
- Orbit differences using both models are small



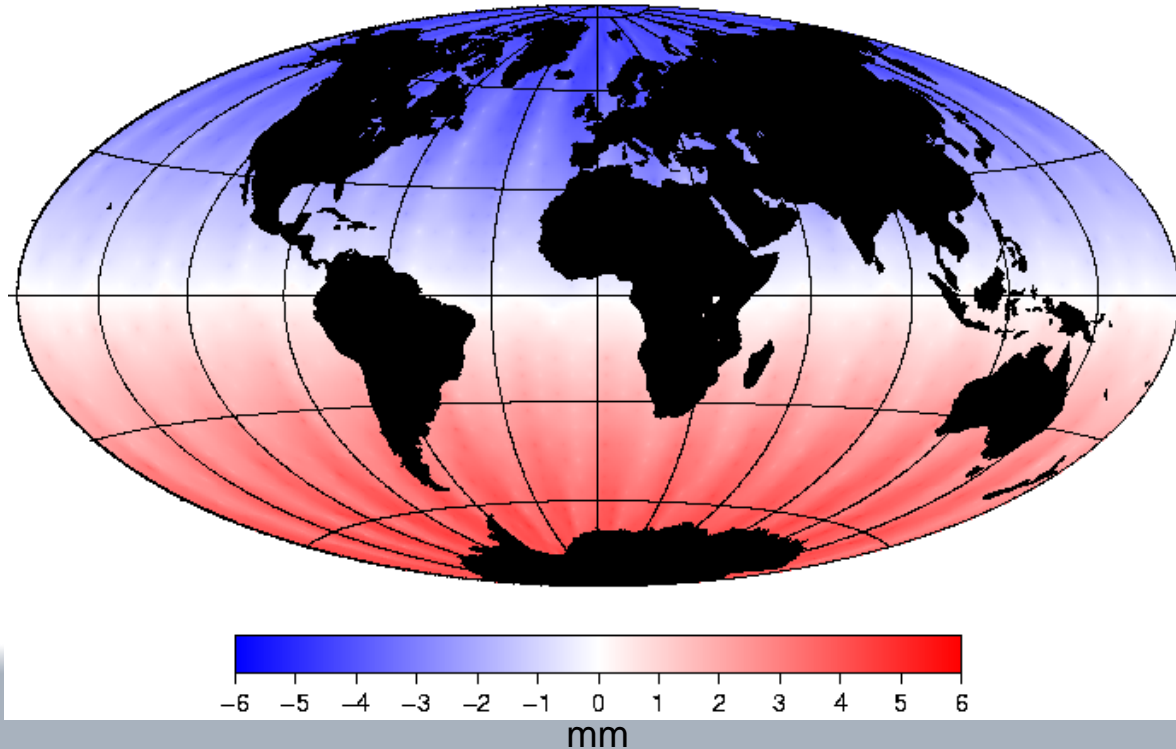
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# ITRF2008 test

RMS of radial difference D+L orbits  
ITRF2005 – ITRF2008

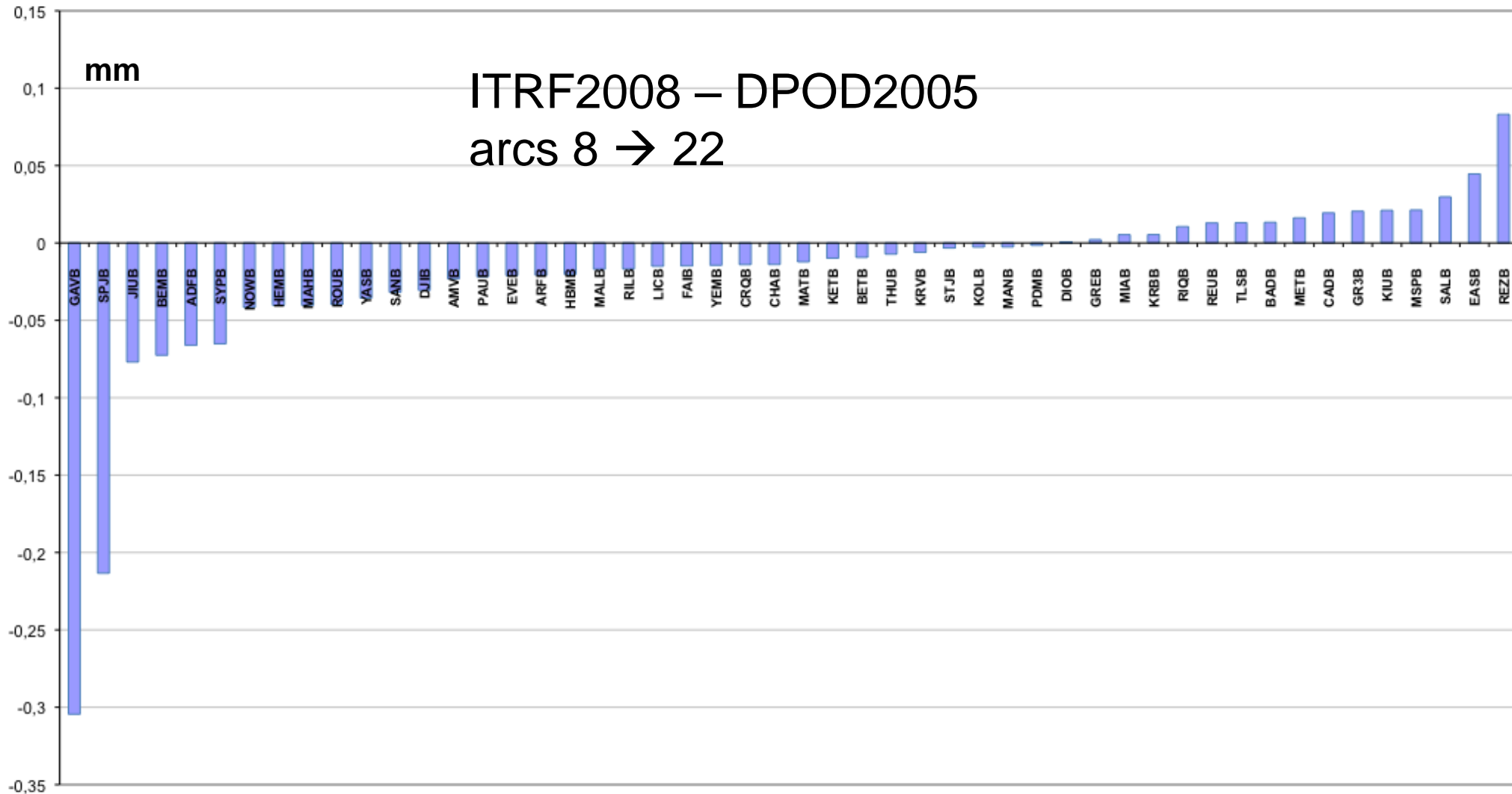


Mean radial difference D+L Orbits  
ITRF2005 – ITRF2008 arcs 008-022

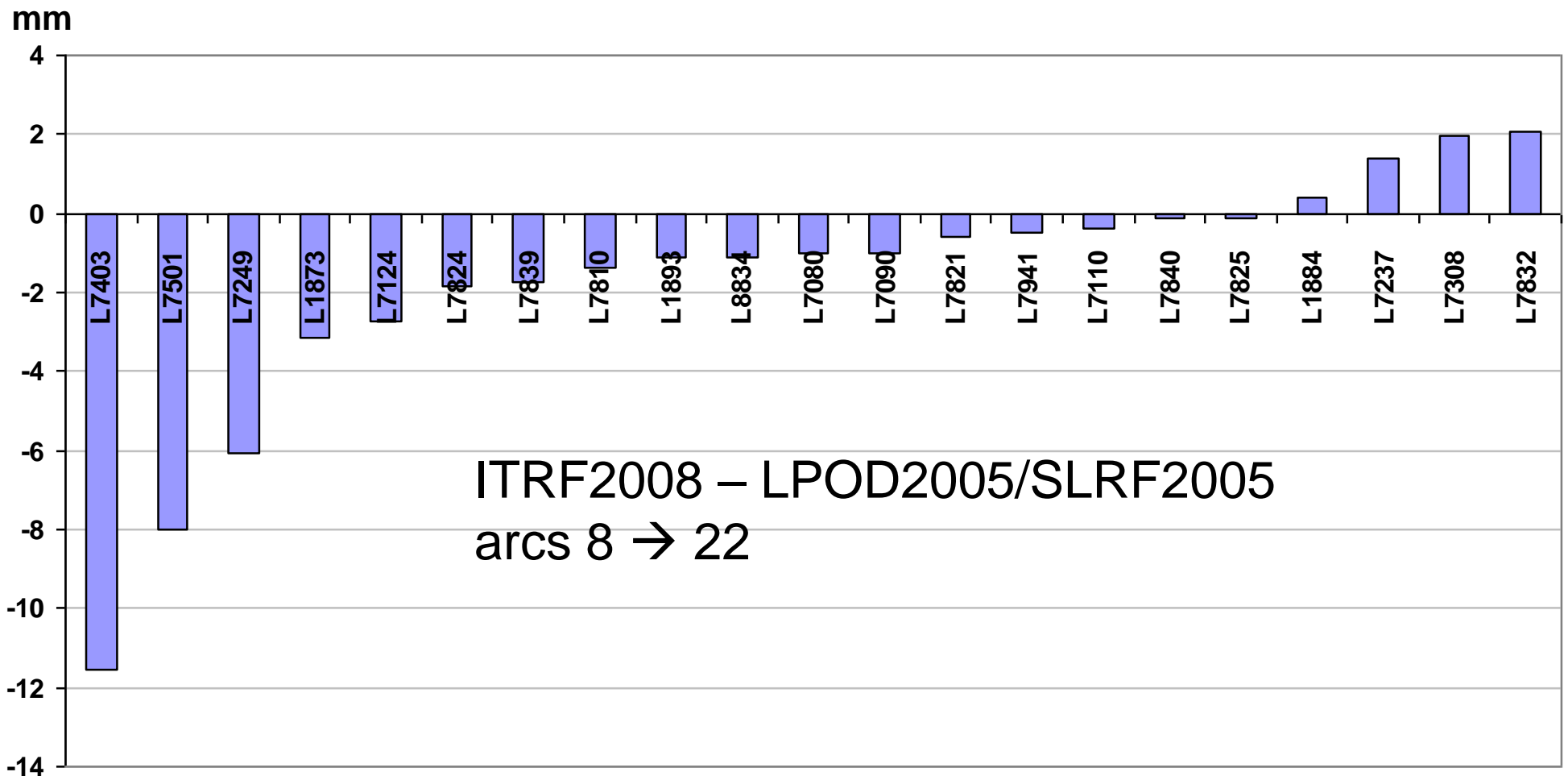


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# RMS of Post fit Doris residuals per station



# RMS of Post fit SLR residuals per station



# Conclusions

- **Radial accuracy of Doris-only orbits as measured by the RMS of SLR residuals is below 2 cm, in line with what is observed on Envisat**
- **Agreement between D+L POE and Doris-only solutions generally below 1 cm**
- **Radial accuracy of POE orbits is in the order of 2 cm**
- **Several modeling improvements have been tested**
  - ◆ **Atmospheric density model (MSIS86 to replace DTM94)**
  - ◆ **Solar radiation pressure (CNES macromodel should be tuned before implementation in GDR orbits)**
  - ◆ **ITRF2008 (Z bias + improvement in both DORIS and SLR residuals)**